
Numerical Methods for ODEs

Sheet 4

Exercise 12: Construction of Adams formulas and BDF methods

- (a) Derive the explicit r -step Adams formulas (Adams-Bashford) for $r = 2, 3$.
- (b) Derive the implicit r -step Adams formulas (Adams-Moulton) for $r = 2, 3$.
- (c) Derive the implicit r -step BDF methods for $r = 2, 3$.

Moreover, think about a construction of these methods for $r > 3$, for instance in Matlab or Maple.

Exercise 13: Consistency of explicit Adams formulas

Show: The explicit Adams formula (Adams-Bashforth) with $r = 2$ has consistency order 2.

Exercise 14: Construction of Nyström- and Milne-Simpson-Methods

In Example 4.5 we discussed further generalizations of multistep methods that can be constructed with the same arguments as in Example 4.3 and Example 4.4.

- (a) Derive a general formula for the construction of Nyström methods. The idea is to integrate the differential equation $y'(t) = f(t, y(t))$ over the interval $[t_{k+r-2}, t_{k+r}]$ of length $2h$. Approximate the right-hand side function by a polynomial that interpolates the function in the points

$$(t_j, f_j) \in \mathbb{R} \times \mathbb{R}^n, \quad j = k, k+1, \dots, k+r-1.$$

- (b) Use the same ideas to construct the Milne-Simpson methods. In this case, we have to interpolate the right-hand side function in the points

$$(t_j, f_j) \in \mathbb{R} \times \mathbb{R}^n, \quad j = k, k+1, \dots, k+r.$$

Compute the coefficients of these methods in case of $r = 1, 2, 3$.

Exercise 15: Proof of Theorem 4.11 (c)

Given is a linear MSM with associated polynomials $\varrho(\mu)$ and $\sigma(\mu)$. Complete the proof of Theorem 4.11 (c), i.e., show that the condition

$$\varphi(\mu) := \varrho(\mu) - \sigma(\mu) \ln \mu \text{ has the root } \mu = 1 \text{ with multiplicity } p + 1.$$

is equivalent to:

The method has at least consistency order p .

Hint: One can substitute $\mu := e^z$ in φ .

Homework 8: Implementation of grid control

- (a) Implement the adaptive grid control strategy from Algorithm 3.58 using the embedded Runge-Kutta method RKF4(5). Use the method of order $p = 4$ for Φ . The method $\hat{\Phi}$ of order $p = 5$ shall be used only for the error estimation. Measure all errors in the Euclidean vector norm.

The proposed syntax of our method is:

```
function [t,y,n_rejected] = RKF45(f,time,y0,h0,options)
```

Here, **n_rejected** is the number of rejected steps, **t** is the time grid, **y** is the approximate solution on the grid points **t**. As usual, **f** is a function handle to the right-hand side, **time** a vector containing the time horizon, **y0** the initial value (column vector) and **h0** the initial grid size. The parameter **options** is an array containing further input parameters:

Key	Symbol	Meaning
alpha_min	$\underline{\alpha}$	lower step size bound
alpha_max	$\bar{\alpha}$	upper step size bound
tol	TOL	error tolerance in each step
rho	ρ	safety factor

Hint: The parameters contained in the Butcher table of RKF4(5) can be generated by the function

```
[a, B, c, chat] = ButcherDiagramm45;
```

which is available on the web page of this lecture.

- (b) Test your implementation using the examples we learned in the lecture (simple pendulum, predator-prey-model, Brusselator model, van-der-Pol oscillator). In the template RKF45.m, the following default values

$$\underline{\alpha} = 0.2, \quad \bar{\alpha} = 5, \quad \text{TOL} = 10^{-6}, \quad \rho = 0.9$$

and the initial grid size $h = 10^{-1}$ are already set. To check your implementation plot the solution as well as the local grid size using:

`plot(t,t(2:end)-t(1:end-1))`

Does the adaptive algorithm work as expected?

- (c) Use your implementation to compute the trajectory of a satellite which is influenced by the gravity of the earth and the moon. The model reads as follows. The earth and the moon move on a circular orbit around the joint center of gravity $(0, 0)$. The initial position of the earth is $(-\mu, 0)$ and of the moon $(1 - \mu, 0)$. Here, $\mu = 0.012277471$ is the ratio between the mass of the moon and of the earth. The trajectory $t \mapsto (u(t), v(t))$ of the satellite can be computed by means of the differential equation

$$\begin{pmatrix} u \\ v \end{pmatrix}'' = \begin{pmatrix} 2v' - \frac{\partial}{\partial u} V(u, v) \\ -2u' - \frac{\partial}{\partial v} V(u, v) \end{pmatrix},$$

where V is a potential defined by

$$V(u, v) = -\frac{1}{2}(u^2 + v^2) + \frac{1 - \mu}{\sqrt{(u + \mu)^2 + v^2}} + \frac{\mu}{\sqrt{(u - \mu + 1)^2 + v^2}}.$$

Note that the potential grows unboundedly when (u, v) tends to the earth or moon. Transform the second-order ODE system to a first-order system. Solve this model with the method implemented in task a). As input data, the values

$$T = 17.0652166, \quad (u, v)(0) = (0.994, 0), \quad (u', v')(0) = (0, -2.001585106)$$

can be used. The corresponding solution is even periodic! A test script (`satellite.m`) is also available on the homepage of this lecture. As an output of this script, you should see a movie illustrating the trajectory of the satellite.

- (d) Try to solve this problem with other ODE solvers that you have implemented before (implicit/explicit Euler, Crank-Nicolson, explicit Runge-Kutta, ...) on equidistant grids. You will probably need a very fine grid.
- (e) Extend the script `satellite.m` by the following functionality. Due to the periodicity of the exact solution, the value $E_h = \|(u_h(0), v_h(0)) - (u_h(T), v_h(T))\|$ can be used to measure the error of the numerical method. Compute and plot the errors for several computations with $TOL \in \{10^{-4}, 10^{-6}, \dots, 10^{-10}\}$. Determine also the experimental convergence rate. What do you observe?

Homework 9: Consistency of multistep methods

Investigate the multistep methods

$$\text{a) } \quad y_{k+2} - \frac{1}{2}y_{k+1} + \frac{1}{4}y_k = \frac{h}{3}(2f_{k+1} + f_k)$$

$$\text{b) } \quad y_{k+2} - y_{k+1} = h(2f_{k+1} - f_k)$$

$$\text{c) } \quad y_{k+1} - y_k = \frac{h}{2}(f_{k+1} + f_k)$$

regarding consistency and determine the consistency order (e.g. using Theorem 4.11). Furthermore, check if these methods are null-stable.